Computational Fluid Dynamics Theory, Numerics, Modelling

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SISO

Winter Term 2017-2018

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- Velocity \vec{u} (3d)

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- Mass conservation
- Momentum conservation
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Physical laws:

- Mass conservation
- Momentum conservation
- Energy conservation
- Equation of state

Example for the equations of state:

$$p = \rho R_s T$$
 and $e = c_{\nu} T$

$$\frac{\mathrm{d}}{\mathrm{d}t} \int_{\Omega(t)} f(x,t) \,\mathrm{d}\Omega = \int_{\Omega(t)} \left\{ \frac{\partial f}{\partial t}(x,t) + \nabla \cdot (f \, \vec{u}) \right\} \,\mathrm{d}\Omega$$

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Look at the mass *m* inside of an arbitrary volume $\Omega(t)$

$$\frac{\mathrm{d}\boldsymbol{m}}{\mathrm{d}\boldsymbol{t}} = \frac{\mathrm{d}}{\mathrm{d}\boldsymbol{t}} \int\limits_{\Omega(\boldsymbol{t})} \rho \; \mathrm{d}\Omega$$

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Continuity equation:

$$rac{\partial
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Force:

$$F = F_{\Omega} + F_{\partial \Omega}$$

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Force:

$$F = F_{\Omega} + F_{\partial \Omega} = \int_{\Omega(t)} \rho \ \vec{f} \ \mathrm{d}\Omega + \int_{\partial \Omega(t)} \underline{\sigma} \ \vec{n} \ \mathrm{d}S$$

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- heat flux: $\int_{\partial\Omega(t)} \kappa \nabla T \cdot \vec{n} \, \mathrm{d}S$

mass conservation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \left(\rho \, \vec{u} \right) = 0$$

mass conservation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \, \vec{u} \,) = 0$$

2 momentum conservation

$$\rho \frac{\partial \vec{u}}{\partial t} + (\rho \, \vec{u} \cdot \nabla) \, \vec{u} = \rho \, \vec{f} \, + \nabla \cdot \underline{\underline{\sigma}}$$

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Implementation @ momentum conservation

$$\rho \frac{\partial \vec{u}}{\partial t} + (\rho \, \vec{u} \cdot \nabla) \, \vec{u} = \rho \, \vec{f} \, + \nabla \cdot \underline{\underline{\sigma}}$$

energy conservation

$$\rho \frac{\partial \mathbf{e}}{\partial t} = \rho \ \mathbf{Q} + \nabla \cdot (\kappa \ \nabla \ \mathbf{T} \) + \nabla \cdot \left(\underline{\sigma} \ \vec{u}\right) - \left(\nabla \cdot \underline{\sigma}\right) \vec{u}$$

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Implementation 2 momentum conservation

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equation of state (e.g. ideal gas equation)

From T = const. with $\frac{d}{dt}\rho = 0$ follows:

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 $\sim \rightarrow$

- **O** Pressure is given with $p \sim \rho$ (equation of state)
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For a newtonian fluid we get the Navier-Stokes equations as Navier-Stokes equations

$$\nabla \cdot \vec{u} = 0 \tag{1}$$

$$\rho \frac{\partial \vec{u}}{\partial t} + \rho \left(\vec{u} \cdot \nabla \right) \ \vec{u} = \rho \ \vec{f} \ - \nabla p \ + \mu \nabla \cdot \underline{\tau}$$
(2)

Note: often, the kinematic viscosity $\nu := \frac{\mu}{\rho}$ is used if $\rho = const$

Dimensionless Navier-Stokes:

Navier-Stokes momentum equation

$$rac{\partial ec{u}}{\partial t} + (ec{u} \cdot
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Define characteristic time T, length L and velocity U with $L = U \cdot T$:

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Dimensionless representation of the momentum equation:

$$\frac{\partial \vec{v}}{\partial \tau} + (\vec{v} \cdot \nabla) \ \vec{v} = \frac{L}{U^2} \vec{f} - \frac{1}{\rho U^2} \nabla p \ + \frac{\mu}{\rho UL} \nabla \cdot \tilde{\underline{\tau}}$$
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dimensionless forcedensity κ
 [⊥] = ^L/_{U²} f
 [−] (look for Froude number)
 pressure rescaling [˜] = ^p/_{ρU²} (NOTE: only for inc. fluid)

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$$Re := \frac{inertia \ forces}{viscous \ forces}$$

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Dimensionless Navier-Stokes equations

$$\nabla \cdot \vec{v} = 0 \tag{3}$$

$$\frac{\partial \vec{v}}{\partial \tau} + (\vec{v} \cdot \nabla) \ \vec{v} = \vec{\kappa} \ - \nabla \vec{p} \ + \frac{1}{Re} \nabla \cdot \tilde{\underline{\tau}}$$
(4)

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 - velocity field perturbations smooth out quickly
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Example: (flow in pipe)

- Reynolds number: $Re = \frac{\rho d v_z}{\mu}$
- Observation: Julius Rotta (at 1950) $Re_{krit.} \approx 2300$



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The smallest scales that influences the turbulent flow by dissipation effects.

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To retain energy conservation at the numerical domain, one have to resolve also the dissipative scales in the Navier-Stokes equation!

The scales are given as: (ϵ is the average dissipation rate)

length:
$$\eta = \left(\frac{\mu^3}{\epsilon \rho^3}\right)^{\frac{1}{4}}$$
 vel: $u_{\eta} = \left(\frac{\mu}{\rho} \epsilon\right)^{\frac{1}{4}}$ time: $\tau_{\eta} = \left(\frac{\mu}{\rho \epsilon}\right)^{\frac{1}{2}}$

with

$$Re_{\eta} = rac{\eta \ u_{\eta} \ \mu}{
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Approximation of the dissipation rate (from large scales):

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Therefore we get the relation:

$$\frac{L}{\eta} = L \cdot \left(\frac{\mu^3}{\epsilon \rho^3}\right)^{-\frac{1}{4}} \sim L \cdot \left(\frac{U^3 \rho^3}{L \mu^3}\right)^{\frac{1}{4}}$$

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Example: $(L \approx 10^3 \text{m}, v \approx 1 \frac{\text{m}}{\text{s}}, \rho \approx 1.3 \frac{\text{kg}}{\text{m}^3}, \mu \approx 17.1 \ \mu \text{Pa} \cdot \text{s})$
$$Re \approx 7.5 \cdot 10^9$$
$$n \approx 4 \cdot 10^{-5} \text{m}$$

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Example: ($L \approx 10^{-3}$ m , $\nu \approx 0.1 \frac{m}{s}$, $\rho \approx 1060 \frac{kg}{m^3}$, $\mu \approx 3 mPa \cdot s$)

$$Re \approx 35$$

 $\eta \approx 7 \cdot 10^{-5} \,\mathrm{m}$

Simulation approaches:



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• Eddy dissipation modelling on small scales:

- Reynolds-Averaged Navier Stokes (RANS)
- Large-Eddy Simulation

• ...

$$oldsymbol{v} = \langle oldsymbol{v}
angle + oldsymbol{v}' \qquad ext{and} \qquad oldsymbol{p} = \langle oldsymbol{p}
angle + oldsymbol{p}'$$

with the mean value $\langle \cdot \rangle$ of \cdot and the fluctuating part \cdot' .

- Special cases: temporal or spatial averaging
- In general: $\langle f(\vec{x},t) \rangle = \lim_{N \to \infty} \sum_{n=1}^{N} f(\vec{x},t)$

• Fluctuating part: $\langle f' \rangle = 0$

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Reynolds equations:

$$abla \cdot \langle ec v
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- In general: $\langle f(\vec{x},t) \rangle = \lim_{N \to \infty} \sum_{n=1}^{N} f(\vec{x},t)$
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$$\nabla \cdot \langle \vec{v} \rangle = 0$$

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 - **(**) $k \epsilon$: good on free flow fields with no walls
 - 2 $k \omega$: near wall approximation is good
 - SST brings the advantage of booth together
Application





Mesh quality determined by:

- area
- aspect ratio
- diagonal ratio
- edge ratio
- skewness
- orthogonal quality
- stretch
- taper
- volume

Mesh - Orthogonal Quality

$$OQ = \min_{i} \left\{ \frac{A_{i}\dot{f}_{i}}{|\vec{A}_{i}||\vec{f}_{i}|}, \frac{A_{i}\dot{c}_{i}}{|\vec{A}_{i}||\vec{c}_{i}|} \right\},$$
(5)

 A_i face normal vector

 f_i vector from the centroid of the cell to the centroid of that face c_i vector from the centroid of the cell to the adjacent cell



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Unacceptable	Bad	Acceptable	Good	Very good	Excellent
0-0.001	0.001-0.14	0.15-0.20	0.20-0.69	0.70-0.95	0.95-1.00

Boundary layer mesh

for flows with high Reynold's number, strong gradients exist within the boundary layer close to a solid wall (with a no-slip boundary condition)

Boundary layer mesh for flows with high Reynold's number, strong gradients exist within the boundary layer close to a solid wall (with a no-slip boundary condition)



Inflation layer examples:





Hints for mesh generation

- minimize mesh complexity
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 - use quad / hex elements when appropriate
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 - use quad / hex elements when appropriate (e.g. boundary layers, long pipes)
- maximize solution accuracy
 - concentrate mesh elements in critical regions (e.g. boundary layers, wakes, shocks)
 - align quad / hex meshes with flow direction
 - avoid poor quality elements (e.g. twisted, skewed)

Application



Problem Definition - Boundary conditions

Choosing appropriate boundary conditions:

- nature of flow incompressible / compressible ...
- physical models turbulence, species transport ...
- position of boundary
- what is known
- convergence of solution may (strongly) depend on choice of boundary conditions

Problem Definition- Numerical solver

two basic solver approaches :

- pressure-based solver
 - originally developed for low-speed flows
 - pressure determined from pressure or pressure-correction equation (obtained from manipulating continuity and momentum equations)
- density-based solver
 - originally developed for high-speed flows
 - density determined from continuity equation
 - pressure determined from equation of state

similar discretization method is used for both pressure-based and density-based solvers.

linearization and solving of the discrete equations is different for two approaches.

Application



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- checking for property conservation
 - overall heat and mass balances should be within 0.1% of net flux through domain

Convergence difficulties

- numerical instabilities can arise due to :
 - ill-posed problem (no physical solution)
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Convergence difficulties

- numerical instabilities can arise due to :
 - ill-posed problem (no physical solution)
 - poor quality mesh
 - inappropriate boundary conditions
 - inappropriate solver settings
 - inappropriate initial conditions
- trouble-shooting
 - ensure problem is physically realizable
 - compute an initial solution with a first-order discretization scheme
 - decrease under-relaxation for equations having convergence problems (segregated)
 - reduce CFL number (unsteady flow)
 - re-mesh or refine mesh regions with high aspect ratio or highly skewed cells

Application



Post Processing

- qualitative analysis (visualization):
 - displaying the mesh
 - contours of flow fields (e.g. pressure, velocity, temperature, concentrations ...)
 - contours of derived field quantities
 - velocity vectors
 - animation (using keyframes or frame-by-frame)
- quantitative analysis:
 - XY plots (e.g. pressure, velocity, temperature vs position)
 - forces and moments on surfaces
 - surface and volume integrals
 - Flow solvers may contain a complete post-processing environment
 - generally not necessary to use external post-processing software